

## 217(10): Comparison of Orbital Linear Velocities

### 1) Focal Orbits

The orbital linear velocity is given by:

$$v^2 = \left(\frac{dr}{dt}\right)^2 + r^2 \left(\frac{d\theta}{dt}\right)^2 \quad - (1)$$

Use:

$$\frac{dr}{dt} = \frac{dr}{d\theta} \frac{d\theta}{dt} \quad - (2)$$

$$\text{so } v^2 = \left(\frac{d\theta}{dt}\right)^2 \left( \left(\frac{dr}{d\theta}\right)^2 + r^2 \right) \quad - (3)$$

The focal orbits are:

$$r = \frac{d}{1 + \epsilon \cos(x\theta)} \quad - (4)$$

$$\text{so } \frac{dr}{d\theta} = \left(\frac{x\epsilon}{d}\right) r^2 \sin(x\theta), \quad - (5)$$

$$\text{and: } v^2 = \left(\frac{d\theta}{dt}\right)^2 r^2 \left( 1 + r^2 \left(\frac{x\epsilon}{d}\right)^2 \sin^2(x\theta) \right) \quad - (6)$$

$$\text{Now use: } \frac{d\theta}{dt} = \frac{L}{mr^2} \quad - (7)$$

$$\text{so } v^2 = \left(\frac{L}{mr}\right)^2 \left( 1 + \left(\frac{x\epsilon r}{d} \sin(x\theta)\right)^2 \right) \quad - (8)$$

so the orbital linear velocity is:

$$v = \frac{L}{mr} \left( 1 + \left( \frac{x(E - E_0) \sin(x\theta)}{d} \right)^2 \right)^{1/2} \quad - (9)$$

where  $r = \frac{d}{1 + \epsilon \cos(x\theta)} \quad - (10)$

$$d = \frac{L^2}{mk}, \quad \epsilon = \left( 1 + \frac{2EL^2}{mk^2} \right)^{1/2} \quad - (11)$$

$$k = mMG.$$

Eq. (9) will give an essentially infinite new subject area in orbital dynamics. For example, if  $x \sim 1$  the precessing orbits of the solar system are obtained.

### Suggested Plots

- 1)  $v$  against  $\theta$  for  $x = 1$  (Newtonian)
- 2)  $v$  against  $\theta$  for  $x \sim 1$  (precessing orbits).

- 3)  $v$  against  $\theta$  for  $x$  increased slightly from  $x \sim 1$  to study the precise way in which the orbit starts to deviate from the precessing ellipse.
- 4)  $v$  against  $\theta$  for  $x$  is general.

The results for EBR are as follows:

(2)

3)

$$\frac{dr}{d\theta} = r^2 \left( \frac{1}{b^2} - \left(1 - \frac{r_0}{r}\right) \left( \frac{1}{a^2} + \frac{1}{r^2} \right) \right)^{1/2} \quad - (12)$$

$$a = \frac{L}{mc}, \quad b = \frac{Lc}{E}, \quad r_0 = \frac{2MG}{c^2}, \quad - (13)$$

$$L = mr^2 \frac{d\theta}{d\tau}; \quad \omega = \frac{cb}{r^2} \left(1 - \frac{r_0}{r}\right), \quad - (14)$$

$$L = mr^2 \omega \frac{d\tau}{dt} \quad - (15)$$

So  $\omega = \frac{L}{mr^2} \frac{d\tau}{dt} = \frac{d\theta}{dt}, \quad - (16)$

where  $\frac{d\tau}{dt} = \frac{mc^2}{E} \left(1 - \frac{r_0}{r}\right) \quad - (17)$

so  $\frac{d\theta}{dt} = \frac{Lc^2}{Er^2} \left(1 - \frac{r_0}{r}\right) \quad - (18)$

The orbital linear velocity ii EGR is therefore:

$$v = \frac{cb}{r} \left(1 - \frac{r_0}{r}\right) \left( r^2 \left( \frac{1}{b^2} - \left(1 - \frac{r_0}{r}\right) \left( \frac{1}{a^2} + \frac{1}{r^2} \right) \right) + 1 \right)^{1/2} \quad - (19)$$

Compare eq. (19) graphically with eq. (9).  
The results will be completely different.