

34(2). Velocity of the Newtonian orbit

The gravitomagnetic field $\underline{\Omega}_g$ is a velocity defined by:

$$\underline{\Omega}_g = \nabla \times \underline{v}_g \quad (1)$$

where \underline{v}_g is a linear velocity. In order to calculate the velocity of an orbit assume that \underline{v}_g is the orbital linear velocity of the orbit. If it is assumed that $\underline{\Omega}_g$ is uniform then:

$$\underline{v}_g = \frac{1}{2} \underline{\Omega}_g \times \underline{r} \quad (2)$$

as in previous notes. It follows that:

$$\begin{aligned} v_g^2 &= \frac{1}{4} (\underline{\Omega}_g \times \underline{r} \cdot \underline{\Omega}_g \times \underline{r}) \quad (3) \\ &= \frac{1}{4} (\Omega_g^2 r^2 - (\underline{\Omega}_g \cdot \underline{r})(\underline{r} \cdot \underline{\Omega}_g)) \end{aligned}$$

If

$$\underline{\Omega}_g \perp \underline{r} \quad (4)$$

then:

$$v_g^2 = \frac{1}{4} \Omega_g^2 r^2 \quad (5)$$

so

$$\Omega_g^2 = 4 \frac{v_g^2}{r^2} \quad (6)$$

For the Newtonian orbit:

$$\Omega_g^2 = \frac{4mb}{r^2} \left(\frac{2}{r} - \frac{1}{a} \right) \quad (7)$$

where a is the semi major axis of the ellipse.
The Newtonian orbit is described by:

$$\frac{1}{r} = \frac{1}{a} (1 + e \cos \theta) \quad (8)$$

2) where α is the half right latitude and e the eccentricity.
In plane polar coordinates the linear velocity is described by:

$$\underline{v}_g = \dot{r} \underline{e}_r + r \dot{\theta} \underline{e}_\theta, \quad - (9)$$

and if T is the time taken for one revolution of the orbit:

$$\Omega_g T = 2\pi \quad - (10)$$

Therefore the vorticity of the orbit is:

$$\begin{aligned} \Omega_g^2 &= \frac{4}{r^2} \left(\left(\frac{dr}{dt} \right)^2 + r^2 \left(\frac{d\theta}{dt} \right)^2 \right) \quad - (11) \\ &= \frac{4 v_g^2}{r^2} \end{aligned}$$

From a Lagrangian analysis:

$$L = m r^2 \frac{d\theta}{dt} \quad - (12)$$

and

$$\frac{dr}{dt} = \frac{dr}{d\theta} \frac{d\theta}{dt} \quad - (13)$$

It follows that:

$$\begin{aligned} \Omega_g^2 &= \frac{4}{r^2} \left(\frac{d\theta}{dt} \right)^2 \left(\left(\frac{dr}{d\theta} \right)^2 + r^2 \right) \quad - (14) \\ &= \frac{4 L^2}{m^2 r^6} \left(\left(\frac{dr}{d\theta} \right)^2 + r^2 \right) \end{aligned}$$

The effect of a precession is to increment the vorticity as follows:

$$\Omega_g^2 \rightarrow \frac{4}{r^2} \left(v_g^2 + v_g^2 (\text{precession}) \right) \quad - (15)$$

So:
$$V_{g1}^2 = \frac{L^2}{m^2 r^4} \left(\left(\frac{dr}{d\theta} \right)^2 + r^2 \right) + \Omega^2 r^2 \quad - (16)$$

where Ω is the observed precession:

$$\Omega = \frac{\Omega_g}{2} \quad - (17)$$

Now define the orbit with precession as:

$$V_{g1}^2 = \frac{L^2}{m^2 r^4} \left(\left(\frac{dr}{d\theta} \right)_1^2 + r^2 \right) \quad - (18)$$

It follows from eqs. (16) to (18) that:

$$\left(\frac{dr}{d\theta} \right)_1^2 = \left(\frac{dr}{d\theta} \right)^2 + \frac{m^2 \Omega^2 r^6}{L^2} \quad - (19)$$

Using eq. (8):

$$\frac{dr}{d\theta} = \frac{\epsilon}{\alpha} r^2 \sin \theta \quad - (20)$$

so
$$\left(\frac{dr}{d\theta} \right)_1^2 = \left(\frac{\epsilon^2}{\alpha^2} \sin^2 \theta \right) r^4 + \left(\frac{m \Omega r^3}{L} \right)^2 \quad - (21)$$

Here:

$$\sin^2 \theta = 1 - \cos^2 \theta \quad - (22)$$

$$= 1 - \frac{1}{\epsilon^2} \left(\frac{d}{r} - 1 \right)^2$$

From these equations it is possible to find the

4) eqn (23):

$$\left(\frac{dr}{d\theta}\right)^2 = r^4 \left(\frac{\epsilon^2}{d^2} \left(1 - \frac{1}{\epsilon^2} \left(\frac{d}{r} - 1 \right)^2 \right) \right) + \left(\frac{m\Omega r^3}{L} \right)^2 - (23)$$

So:

$$d\theta = \left(r^4 \left(\frac{\epsilon^2}{d^2} \left(1 - \frac{1}{\epsilon^2} \left(\frac{d}{r} - 1 \right)^2 \right) \right) + \left(\frac{m\Omega r^3}{L} \right)^2 \right)^{-1/2} dr - (24)$$

It is possible to try to numerically integrate eq. (24) to give θ as a function of r for the new orbit.
